

National Aeronautics and Space Administration



### Silicon Micro Dosimeter for High-Altitude Measurements of Cosmic Radiation

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# Background & Motivation Dosimeter Requirement Measurement Data Future Work & Improvements



# **RaD-X Science Motivation**

#### NASA Strategic Goal:

Improve prediction of biologically harmful radiation at aviation altitudes



**RaD-X:** Important step in the V&V process en route to making NAIRAS operational





# **RaD-X Science Goals**



- 1. Improve tools that predict energy deposition characteristics of penetrating GCR in Earth's atmosphere
  - Combine different dosimeter measurements and two flight altitudes to assess model uncertainty in GCR primaries
- 2. Identify and characterize low-cost radiation measurement solutions Continuous, global measurements for real-time data assimilative modeling



Image Credit: NASA LaRC

# **Project Overview**



#### **Flight Overview**

- Launched Sept. 25<sup>th</sup>, 2015
- Over 18 hours of dosimetric data above 20 km altitude

#### **Science Instruments**

- Teledyne UDOS001 micro dosimeter (TID)
- FarWest Technologies tissue equivalent proportional counter (TEPC)
- Bulgarian Academy of Sciences Liulin linear energy transfer (LET) spectrometer
- University of Surrey RaySure dosimeter







# 1. Background & Motivation

# 2. Dosimeter Requirements

# 3. Measurement Data

# 4. Future Work & Improvements



# **TEPC Reference Dosimeter**



#### **TEPC Specifications**

| Parameter                       | Value(s)                           |
|---------------------------------|------------------------------------|
| Total Mass                      | 4 kg                               |
| Power                           | 0.5 W                              |
| Dosimeter outer<br>dimensions   | 160 mm diameter x<br>340 mm height |
| Detector size<br>(spherical)    | 12.7 cm diameter                   |
| A150 detector wall<br>thickness | 0.21 cm                            |



# **TID Silicon Dosimeter**



# TEPC

TID

#### **TID Sensor Specifications**

| Parameter                      | Value(s)         |  |
|--------------------------------|------------------|--|
| Measured quantity              | Accumulated dose |  |
| Total Mass                     | 20 g             |  |
| Power                          | 0.28 W           |  |
| Dosimeter outer<br>dimensions  | 35 x 25 x 4.5 mm |  |
| Silicon detector<br>dimensions | 5 x 5 x 0.25 mm  |  |
| DAC low range                  | 0 – 3.5 mRads    |  |
| DAC low step size              | 13.6 µRads       |  |
| DAC low step size              | 19.5 mV          |  |
| DAC logarithmic range          | 0 – 68 kRads     |  |







#### **Flight Heritage**

- NASA's Lunar Reconnaissance Orbiter [Mazur et al., 2011]
- DC-8 aircraft for the Automated Radiation Measurements for Aerospace Safety (ARMAS) project [Tobiska et al., 2011]

#### **Radiation Beam Characterization**

- [Lindstrom et al., 2011]
- [Straume et al., 2016]

#### **TID Sensor Specifications**

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# **TID Interface Design**



#### Requirements

- **Power:** 13 to 40 VDC at 10 mA
- **Data:** Four analog channels, 0-5V output, low impedance to ADC (TID sensor has ~ 10 k $\Omega$  Zo)
- **Control**: On/Off switching capability ٠
- Mechanical: Survive ~15g, view of the ٠ sky



#### **Voltage Buffers**





# Background & Motivation Dosimeter Requirements

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# Laboratory measurements with flight hardware showed good agreement with theory



Not shown: 5 cm piece of L200 minicell polyethylene to attenuate secondary electrons generated in the air gap





#### Observed < 10 mV peak-to-peak noise on TID analog outputs



Zoomed TID Laboratory Measurements (Low and Med. channels)

3/4/2018

attenuate secondary electrons generated in the air gap





TID flight data showing the Low and Medium channels Average dose rate of roughly 3.15  $\mu Gy/hr$  for the entire mission







Flight data showed significant noise injected by a different instrument (~150 mV peak-to-peak)



#### **Noise Source**

- Upstream DC-DC converter used for a different instrument
- Noise identified during EMI/EMC test
  - Project decided to accept the risk without modification





#### TID output steps are still clearly visible in the noisy data



#### TID measurements from flight (Low channel)

#### **Post-Processing**

- Convert voltage to dose (µGy)
- 5-min average for each data sample
- Convert to instantaneous dose rate (µGy/hr)







| Barometri<br>c Altitude<br>(km) | TEPC<br>Reference<br>Dosimeter Dose<br>Rate (uGy/hr) | TID Dose<br>Rate<br>(uGY/hr) |
|---------------------------------|--|------------------------------|
| 24.6                            | $3.05\pm0.48$  | $3.52\pm0.70$                |
| 36 <sub>Figure an</sub>         | d table fr58 time tens, et a                         | $1.,2055 \pm 0.51$           |





# Background & Motivation Dosimeter Requirements Measurement Data Conclusions & Future Work





Silicon Dosimeters for Quantifying Biologically Harmful Radiation

- TID measurements are consistent with wellvalidated and verified dosimeter (TEPC)
- Accurate empirical calibration between the TID's measured absorbed dose in silicon and effective dose are needed





**NAIRAS Findings** 

- Model was missing pion-initiated electromagnetic cascade processes
- Preliminary results with updated transport algorithm show significant improvement
  - More extensive validation is needed

# Future Work



- Additional high-altitude balloon flights with the RaD-X payload
- Revised power distribution layout



## Questions



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Image Credit: NASA LaRC





- Four instruments were characterized at LLNL
  - Cobalt-60 gamma ray source (NIST-traceable)
  - Californium-252 fission radiation neutron and gamma ray source (NIST-traceable)
- Provided a functional test for the TID sensor
  Did not simulate RaD-X radiation environment
- More information: [Straume et al., 2016]

# **Payload Avionics**



#### All avionics are COTS with exception of PC/104 Interface Circuit, TID Interface Circuit, and most cables

- Flowed down interfaces and requirements to the subsystem level
- Analyzed power and data interfaces to ensure correct part selection
- Collaborated with other subsystem leads to ensure payload level requirements were met

# Payload Overview





All avionics are COTS with exception of PC/104 Interface Circuit, TID Interface Circuit, and cables



# RaD-X Flight Track



Launch 1:15 PM EDT Ascent to Region B

**NASA** 



5 November 2015



# RaD-X Team



